

APPROVAL CERTIFICATE

EASA.21J.361

Pursuant to Regulations (EC) 216/2008 and (EU) 748/2012 and
subject to the conditions specified below, the Agency hereby
certifies

TC INTER - INFORMATICS a.s.

Kolcavka 75/3

P.O. BOX 5

190 00 Praha 9

Czech Republic

as a DESIGN ORGANISATION

approved according to Part 21, Section A, Subpart J.

CONDITIONS :

1. The approval is limited to that specified in the enclosed Terms of Approval, and
2. This approval requires compliance with the procedures specified in the Design Organisation Handbook, reference P02 Příručka projekční organizace, in the latest revision, and
3. This approval is valid whilst the approved Design Organisation remains in compliance with Part 21, Section A, Subpart J.
4. Subject to compliance with the foregoing conditions, this approval shall remain valid until surrendered or revoked.

For the **European Aviation Safety Agency**,

Date of issue: 21 July 2015

Dominique ROLAND

Head of Design Organisations Department



Terms of Approval

Design Organisation Approval Certificate

EASA.21J.361

1 Scope of approval

This Design Organisation Approval has been granted for:

- designing in accordance with the applicable type-certification basis and environmental protection requirements changes and repairs to aircraft in the following areas:
 - cabin interiors
 - structure
 - environmental systems
 - electrical systems
 - installation of avionics equipment
 - hydro mechanical systems
 - powerplant and fuel systems
 - galleys and other interiors equipment
 - non-critical engine parts
- showing and verifying the compliance with the applicable type-certification basis and environmental protection requirements, and
- demonstrating to the Agency this compliance.

2 Categories of products

Aeroplanes and Rotorcraft

3 List of products

[Not applicable]

4 Privileges

- a) The holder of this design organisation approval shall be entitled to perform design activities under Part21 and within its scope of approval.
- b) Subject to 21.A.257(b), the Agency shall accept without further verification compliance documents submitted by the holder of this design organisation approval for the purpose of obtaining:
 - 1. the approval of flight conditions required for a permit to fly; or
 - 2. a supplemental type-certificate; or
 - 3. a major repair design approval.
- c) The holder of this design organisation approval shall be entitled, within its terms of approval and under the relevant procedures of the design assurance system:

1. to classify changes to type design and repairs as "major" or "minor";
2. to approve minor changes to type design and minor repairs;
3. to issue information or instructions containing the following statement: "The technical content of this document is approved under the authority of DOA ref. EASA.21J.361";
4. to approve Minor Revisions to Aircraft Flight Manuals and issue such revisions containing the following statement: 'Revision No [YY] to AFM (or supplement) ref. [ZZ] is approved under the authority of DOA ref. EASA. 21J.361'.
5. to approve the design of major repairs to products for which it holds the type-certificate or the supplemental type-certificate
6. to approve the conditions under which a permit to fly can be issued in accordance with 21A.710(a)(2),
 - (i) except for initial flights of:
 - a new type of aircraft, or
 - an aircraft modified by a change that is or would be classified as a significant major change, or
 - an aircraft whose flight and/or piloting characteristics may have been significantly modified.
 - (ii) except for permits to fly to be issued for the purpose of 21A.701(a)(15).
7. to issue a permit to fly in accordance with 21A.711(b) for an aircraft it has designed or modified, and when the design organisation itself is controlling under its DOA the configuration of the aircraft and is attesting conformity with the design conditions approved for the flight.

5 Limitations

[See in 1, above]

1. Structure limited to secondary structure and simple modifications of primary structure components.
2. Environmental systems limited to installation of components of the system and design of tubing and similar components.
3. Hydro mechanical systems limited to installation of components of the systems and design of piping and similar components.
4. Electrical systems limited to electrical subsystem integration and installation of electrical harness design and installation.
5. Powerplant and fuel systems limited to fuel system installation of components and design of tubing and similar components.
6. Privilege to approve the flight conditions supporting permits to fly, within technical capability defined in the scope.

Date of issue: 21 July 2015

Dominique ROLAND

Head of Design Organisations Department